

# Analysis of the Influence of the Thickness and the Hole Radius on the Calibration Coefficients in the Hole-Drilling Method for the Determination of Non-uniform Residual Stresses

J.M. Alegre<sup>1</sup> · A. Díaz<sup>1</sup> · I.I. Cuesta<sup>1</sup> · J.M. Manso<sup>2</sup>

Received: 15 May 2018 / Accepted: 16 September 2018

© Society for Experimental Mechanics 2018

## Abstract

The Hole-Drilling method is a semi-destructive technique useful for obtaining residual stress distributions by drilling and measuring relieved strains. The standard for this method, i.e., ASTM E837 – 13a, is based on the Integral Method and facilitates obtaining the coefficient matrices required to solve the inverse problem and to calculate the residual stress at depths of up to 1.00 mm. A possible deviation from the coefficients given by this standard is searched when the piece has a small thickness or the hole diameter is not 2.00 mm. FEM simulations are performed with the aim of analysing these effects and proposing new matrices, expressions and correlations for conditions outside the usual thickness and diameter limits. A parametric sweep over a wide range of thicknesses and hole diameters has been implemented in ANSYS to establish a consistent and automated numerical procedure for widening the applicability of the Hole-Drilling method.

**Keywords** Hole-drilling · Residual stress · Finite element method · Strain gauges

## Introduction

Residual stresses are present in many industrial components due to fabrication processes. For example, welding induces thermal expansion near the heat-affected zone and thereby introduces residual stress, which can promote unexpected failure [1, 2]; machining or cold-forming can also generate residual stresses [3] that affect fatigue life [4]. However, sometimes, a compressive residual stress distribution is introduced with the objective of improving the fatigue life [5–7]. Therefore, knowing the actual stress state of a piece is fundamental to evaluating its integrity during its service life. Measurement techniques can be classified into destructive and non-destructive methods [8]. Although the Hole-Drilling method removes material from the target piece, the hole is usually so small in comparison with the piece's dimensions that the component's

integrity is almost unaffected; such measurements may be regarded as an intermediate situation between destructive and non-destructive, i.e., semi-destructive [8].

The Hole-Drilling method comprises three stages: drilling operation, registration of relieved strains and stress calculation. The relieved strains might be measured using optical methods, such as Digital Image Correlation or interferometry [9]. However, those methods have not been standardised in the Hole-Drilling framework; instead, strain gauge rosettes are considered in ASTM E837 – 13a [10]. Here, rosettes comprising three strain gauges are analysed, but it should be noted that other common configurations include a four-gauge rosette (for plasticity correction [11–13]) and a special-purpose six-gauge rosette (for measurements requiring high sensitivity [14]). Rosettes of types A and B are shown in Fig. 1. Both have two strain gauges forming a 90° angle, but while rosette A has the third gauge on the prolongation of their bisector, i.e., in the opposite quadrant, in rosette B, this third gauge forms a 45° angle with the other two, i.e., it is in the same quadrant.

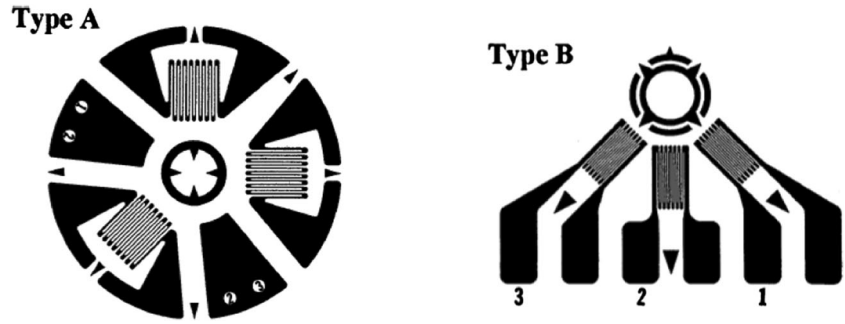
The rosette dimensions are standardised, giving three independent parameters: the rosette diameter  $D$ , the gauge length  $GL$  and the gauge width  $GW$ . The gauge width of a type-B rosette must be less than the  $GW$  of a type-A rosette because three strain gauges are placed in the same quadrant. In addition, radii comprising the rosette are

✉ A. Díaz  
adportugal@ubu.es

<sup>1</sup> Structural Integrity Group, Escuela Politécnica Superior, Universidad de Burgos, Av Cantabria s/n, 09006 Burgos, Spain

<sup>2</sup> SUCONS Research Group, Escuela Politécnica Superior, Universidad de Burgos, C/Villadiego s/n, 09001 Burgos, Spain

Fig. 1 Types of 3-strain-gauge rosettes [10]



65 usually defined, especially for node placement in Finite  
66 Element simulations:

$$69 \quad R_1 = \frac{1}{2}(D - GL) \quad (1)$$

$$68 \quad R_2 = \frac{1}{2}(D + GL) \quad (2)$$

70 All the geometric parameters are graphically defined in  
74 Fig. 4. This paper focuses on the last stage of the Hole-  
75 Drilling procedure, i.e., the numerical relationship between  
76 the relieved strains and the actual residual stress distribu-  
77 tion. There are several calculation algorithms whose objec-  
78 tive is to relate the relieved strains and the residual stress  
79 [15]. ASTM E837 – 13a standardises the integral approach  
80 and establishes a procedure for obtaining the non-uniform  
81 residual stresses in the first millimetre of depth. Following  
82 the standard, a minimum thickness is required for a work-  
83 piece to be considered “thick” [10] and for the non-uniform  
84 stress calibration coefficients to be valid.

85 The Integral Method is a mathematical approach based on  
86 the superposition principle [16]; this algorithm is reviewed in  
87 “Results and Discussion” section. The main objective of this  
88 study is to discuss the procedure found in ASTM E837 – 13a  
89 and to widen its range of applicability. Matrices of calibration  
90 coefficients are re-evaluated to consider the influences of the  
91 thickness and the hole diameter. All calculations are made  
92 using Finite Element simulations, and an automated procedure  
93 for parametric evaluation is established.

## 94 Integral Method

95 The relieved strain registered at drilling depth  $h$  depends on  
96 the stress at depth  $\sigma(z)$  and can be found by integrating over  
97 every depth increment  $dz$  [17]:

$$\varepsilon(h) = \frac{1 + \nu}{E} \int_0^h G(z, h) \sigma(z) dz \quad (3)$$

98 where  $G(z, h)$  are so-called kernel functions, which quantify  
99 the sensitivity of the measured strain to the stress at depth  $z$ ; 100  
101 the deeper the stress applied is, the lower its influence on the  
102 relieved strains measured at the surface. Equation (3) consti-  
103 tutes an inverse problem; therefore, obtaining the stress distri-  
104 bution is not straightforward. The Integral Method is based on  
105 transforming the continuous problem into a set of discrete  
106 equations. Assuming a three-strain-gauge rosette, the follow-  
107 ing three combinations of strains are usually defined [10]:

$$p = \frac{\varepsilon_3 + \varepsilon_1}{2} \quad q = \frac{\varepsilon_3 - \varepsilon_1}{2} \quad t = \frac{\varepsilon_3 + \varepsilon_1 - 2\varepsilon_2}{2} \quad (4)$$

110 Because the registered strain evolves, these combinations 110  
111 are actually functions of depth or vectors whose components 111  
112 correspond to each step during material removal. The corre- 112  
113 sponding stress combinations are: 113  
114

$$P = \frac{\sigma_3 + \sigma_1}{2} \quad Q = \frac{\sigma_3 - \sigma_1}{2} \quad T = \tau_{13} \quad (5)$$

115 An integral equation similar to (3) that includes the kernel 115  
116 functions  $A(z, h)$  and  $B(z, h)$  is defined for each stress-strain 116  
117 pair: 117

$$p(h) = \frac{1 + \nu}{E} \int_0^h A(z, h) P(z) dz \quad (6) \quad 123$$

$$q(h) = \frac{1}{E} \int_0^h B(z, h) Q(z) dz \quad (7) \quad 122 \quad 126$$

$$t(h) = \frac{1}{E} \int_0^h B(z, h) T(z) dz \quad (8) \quad 124 \quad 129$$

128 To handle this set of inverse problems, the equations are 128  
129 transformed into a discrete matrix system in which  $p_i, q_i$  and  $t_i$  129  
130 are the strain combinations obtained experimentally for each 130  
131 depth increment  $i$  [10]: 131  
132  
133

$$p_i = \frac{1 + \nu}{E} \sum_{j=1}^{j=i} \bar{a}_{ij} P_j \quad (9)$$

138  $q_i = \frac{1}{E} \sum_{j=1}^{j=i} \bar{b}_{ij} Q_j$  (10)

139  $t_i = \frac{1}{E} \sum_{j=1}^{j=i} \bar{b}_{ij} T_j$  (11)

140 By solving these three matrix systems, the unknown vectors  
 141  $P_j$ ,  $Q_j$  and  $T_j$  are found; finally, using the equations in (5), the  
 142 stress distribution can be obtained for each depth  $j$  in the coordi-  
 143 nate system defined by the gauges. The non-dimensional co-  
 144 efficients  $\bar{a}_{ij}$  and  $\bar{b}_{ij}$  represent the strain relaxation after removal  
 145 step  $i$  due to unit stress at depth  $j$ . They are related to the kernel  
 146 functions through the following expressions:

147  $\bar{a}_{ij} = \int_{z_{j-1}}^{z_j} A(z, h_i) dz$  (12)

148  $\bar{b}_{ij} = \int_{z_{j-1}}^{z_j} B(z, h_i) dz$  (13)

149 In the present study, the strain relaxation matrices are ex-  
 150 amined using FEM simulations. However, sometimes, the cumu-  
 151 lative strain relaxation coefficients are found instead. In the  
 152 equi-biaxial situation (i.e.,  $p_i$ ,  $P_j$ ) the difference in physical  
 153 meaning between  $\bar{a}_{ij}$  and  $\bar{A}_{ij}$  is shown in Fig. 2. The cumula-  
 154 tive strain relaxation coefficients are defined by [18]:

155  $\bar{A}_{ij} = \int_0^{z_j} A(z, h_i) dz$  (14)

156  $\bar{B}_{ij} = \int_0^{z_j} B(z, h_i) dz$  (15)

167 **Range of Applicability of ASTM E837 – 13a**

168 Non-uniform stresses can only be calculated for “thick” pieces  
 169 according to the ASTM standard. The recommended work-  
 170 piece thickness, hole diameter and depth step size are shown  
 171 in Table 1. These values correspond to the most commonly  
 172 used rosettes, types A and B, with a diameter of 5.13 mm:

**Table 1** Recommendations for workpiece thickness, hole diameter and depth step size in ASTM E837-13a [10]

Max. thickness for a “thin workpiece”	Uniform stress		
	Min. $D_0$	Max. $D_0$	Step size
1.03	1.52	2.54	0.10
Min. thickness for a “thick workpiece”	Non-uniform stress		
	Min. $D_0$	Max. $D_0$	Step size
5.13	1.88	2.12	0.05

To sum up, when applying the standard the following aspects must be considered:

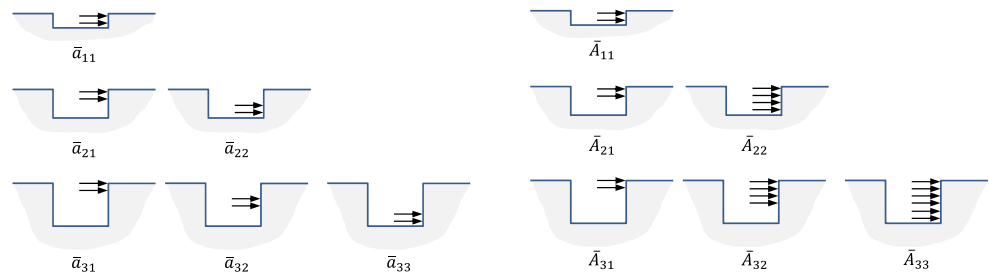
- Residual stresses can only be calculated in a 1 mm layer.
- Non-uniform stresses can only be calculated for pieces that are more than 5.13 mm thick.
- Plasticity effects are not included because this standard is based on the superposition principle.

In addition, the  $\bar{a}_{ij}$  and  $\bar{b}_{ij}$  matrices are tabulated only for  $D_0 = 2$  mm, and a correction factor is introduced for other hole diameters. Here, possible deviations from this correction are evaluated. A discussion of the limitations associated with plasticity is beyond the scope of the present paper; information regarding plastic correction can be found in other studies [12, 20].

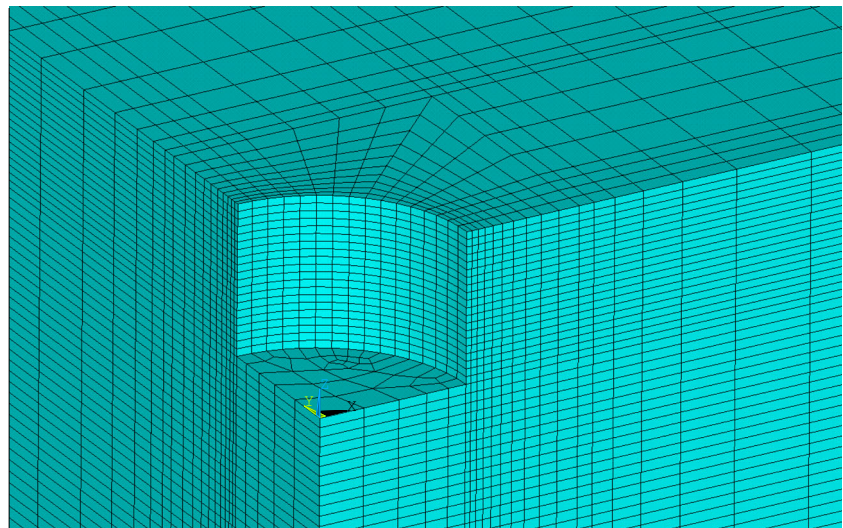
**Finite Element Procedure and Validation**

The Finite Element Method has already been used by many authors for determining  $\bar{a}_{ij}$  and  $\bar{b}_{ij}$ , the coefficients of the calibration matrices [15, 21–25]. The well-known finite elements code ANSYS is used, and a three-dimensional model is created. As a consequence of the symmetry conditions, only a quarter of the model is necessary. An example of the mesh developed is shown in Fig. 3. An element size of 0.05 mm is used in the inner surface of the simulated drilled hole where the loads are applied. Distributed stresses have been applied in

**Fig. 2** Physical meaning of the strain relaxation and cumulative strain relaxation coefficients (adapted from [19])



**Fig. 3** Simulated drilled hole: Finite Element geometry and mesh



196 0.05-mm increments, following [15]; finer mesh was investi-  
 197 gated but results were practically unaffected so the element  
 198 size of 0.05 mm is considered for all simulations. A minimum  
 199 of 16 elements in the hoop direction is used to model a quarter  
 200 of the workpiece. The element type selected is SOLID185,  
 201 which is commonly used for three-dimensional modelling of  
 202 solid structures. This element type features plasticity, stress  
 203 stiffening, large deflection, and large strain capabilities.  
 204 Although in this study, only a linear elastic analysis is re-  
 205 quired, the model can be used to consider non-linear effects.

206 The main fixed dimensions used for the workpiece simu-  
 207 lation are initial diameter of the gauge circle  $D = 5.13 \text{ mm}$ ,  
 208 gauge length  $GL = 1.59 \text{ mm}$ , and external dimension of the  
 209 workpiece  $D_{\text{max}} = 12 \text{ mm}$ . The thickness of the workpiece  
 210 and the diameter of the drilled hole are modified for each  
 211 simulation with the objective of performing a parametric  
 212 study. The dimensions are illustrated in Fig. 4.

213 The strain measured by the strain gauges is obtained from  
 214 the radial displacement of the initial and final points of the  
 215 middle axis of the strain gauge, according to  $[u_x(R_2) - u_x(R_1)] /$   
 216  $GL$ . It must be noted that this measured deformation consti-  
 217 tutes a simplification because the gauge width effect is  
 218 neglected. Additionally, only stresses on the x-y plane have  
 219 been considered; stresses at the hole's bottom face have been  
 220 neglected, though they can appear due to material removal.

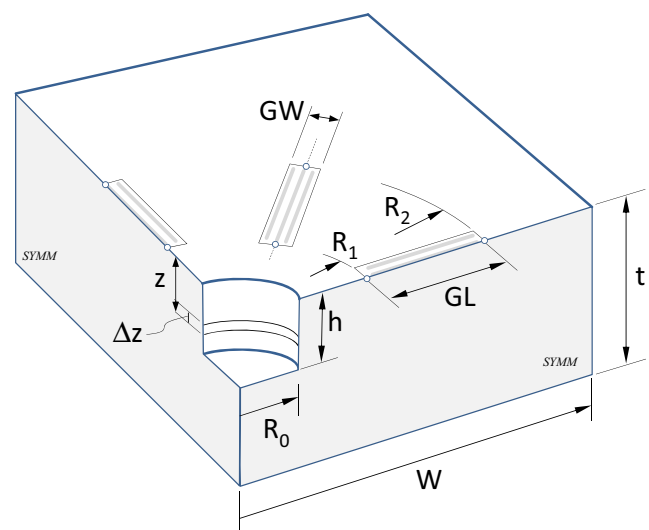
221 However, the validation presented above demonstrates  
 222 that the approximation is good enough for calculating the  
 223 strain based on the strain gauge measurement.

224 An initial validation analysis is performed. For this pur-  
 225 pose, the coefficients obtained for the isotropic stresses  $\bar{a}_{ij}$   
 226 and for the shear stresses  $\bar{b}_{ij}$  are compared with those provided  
 227 in ASTM E837 – 13a. The obtained Hole-Drilling calibration  
 228 coefficients in the matrix,  $\bar{a}_{ij}$  and  $\bar{b}_{ij}$ , are presented in Figs. 5

229 and 6, respectively, together with the calibration coefficients  
 230 collected in ASTM E837 – 13a that correspond to a type-B  
 231 rosette. That type of rosette is chosen for validation because its  
 232 gauge width is smaller than that of a type-A rosette; therefore,  
 233 behaviour that is more similar to the present FE simulations is  
 234 expected. Very good agreement between the coefficients is  
 235 observed. The results corresponding to intermediate hole  
 236 depths ( $h = 0.20 \text{ mm}$ ,  $0.25 \text{ mm}$ , etc.) are omitted for clarity.

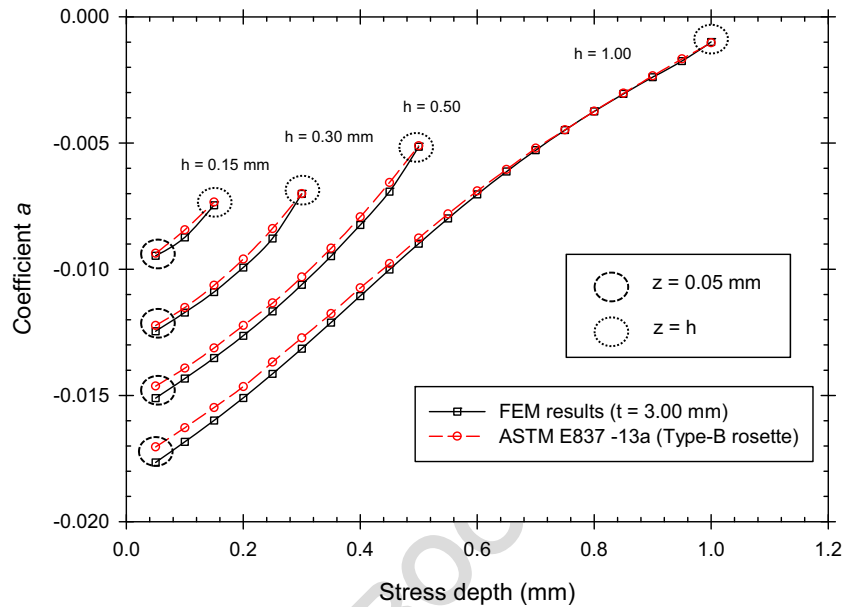
## Results and Discussion

237  
 238 With the aim of assessing the effects of the thickness and the hole  
 239 radius, a parametric study was conducted. Considering the values



**Fig. 4** Main dimensions of the simulated geometry (not to scale)

**Fig. 5** Comparison between matrix  $\bar{a}_{ij}$  for a type-B rosette (from Table 6(a) in ASTM E837-13a) and the FE simulation with  $R_0 = 1.00$  mm and thickness = 3.00 mm



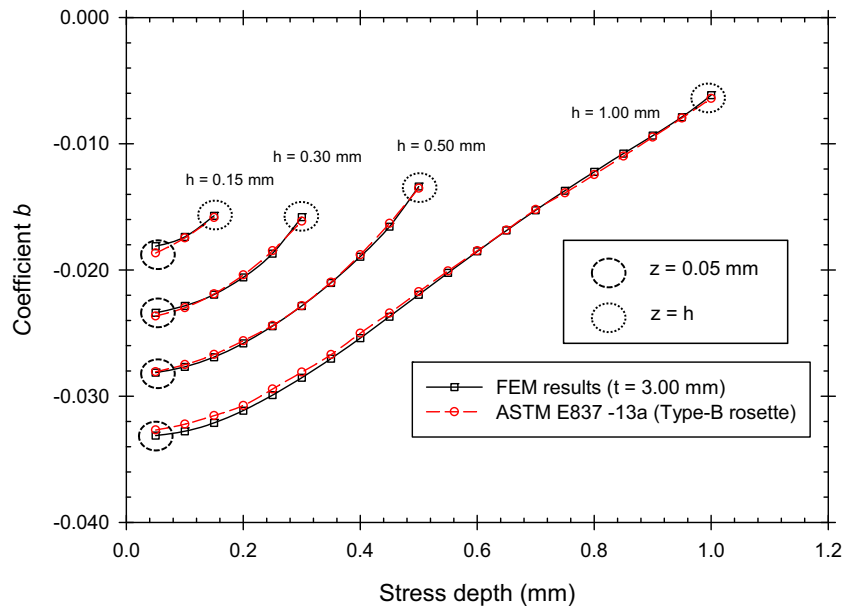
240 shown in Table 2, a parametric sweep over every possible combination was performed, giving a total of 40 simulations.

241  
242 The results corresponding to a 1.00 mm hole radius and a 3.00 mm thickness have already been shown for validation purposes. In the following sections, even though 40 complete matrices were obtained, only the most substantial results are presented. Selected complete matrices for small thicknesses are given in Appendix 1, as the objective of this paper is to extend the applicability of ASTM E837-13a using the matrix Integral Method.

**Effect of the Thickness**

250  
251 For a fixed hole radius of 1.00 mm, as in ASTM E837-13a, the effect of the thickness on the calibration coefficients was evaluated. The graph for all the simulated thicknesses shown in Fig. 7 is for the row corresponding to a hole depth of 1.00 mm, i.e., the last row of the  $\bar{a}_{ij}$  matrix, which demonstrates that the trend is asymptotic. This result explains why the standard establishes a minimum workpiece thickness, which is equal to a rosette diameter of 5.13 mm,

**Fig. 6** Comparison between matrix  $\bar{b}_{ij}$  for a type-B rosette (from Table 6(a) in ASTM E837-13a) and the FE simulation with  $R_0 = 1.00$  mm and thickness = 3.00 mm



**Table 2** Simulated values of the hole radius and the workpiece thickness

Parameter	Range evaluated
Hole radius: $R_0$ [mm]	0.25, 0.50, 0.75, 1.00, 1.25
Thickness: $t$ [mm]	1.05, 1.10, 1.25, 1.50, 2.00, 3.00, 5.00, 10.00

259 as shown in Fig. 7. It is worth noting that for small thick-  
 260 nesses, two different curvatures are obtained in the evolu-  
 261 tion of the effect of the thickness; they depend on whether  
 262 the coefficient corresponds to a stress applied near the bot-  
 263 tom of the hole (top lines, near  $z = 1.00$  mm) or next to the  
 264 workpiece's surface (bottom lines, near  $z = 0.05$  mm). This  
 265 shape of  $\bar{a}_{ij}$  Fig. 7 is caused by the localised bending [26]  
 266 and the change in behaviour is delimited by the coefficient  
 267 corresponding to  $z = 0.60$  mm, approximately. This change  
 268 is attributed to the influence of the free surface at the bot-  
 269 tom: for a very low thickness, the constriction associated  
 270 rwith the bottom is very weak and the deformation increas-  
 271 es to a negative value when stress is applied near  
 272  $z = 0.05$  mm (more compression in the strain gauge sur-  
 273 face) and even to a positive value when stress is applied  
 274 near  $z = 1.00$  mm (traction at the the surface to which the  
 275 straing gauge is applied).

276 For the coefficient line corresponding to  $z = 0.05$  mm, a  
 277 similar tendency is observed for the values of the  $\bar{b}_{ij}$  matrix:  
 278 compressive deformation increases for low thicknesses be-  
 279 cause the hole is less constrained. However, due to the  
 280 association of the  $\bar{b}_{ij}$  matrix with a shear stress state, posi-

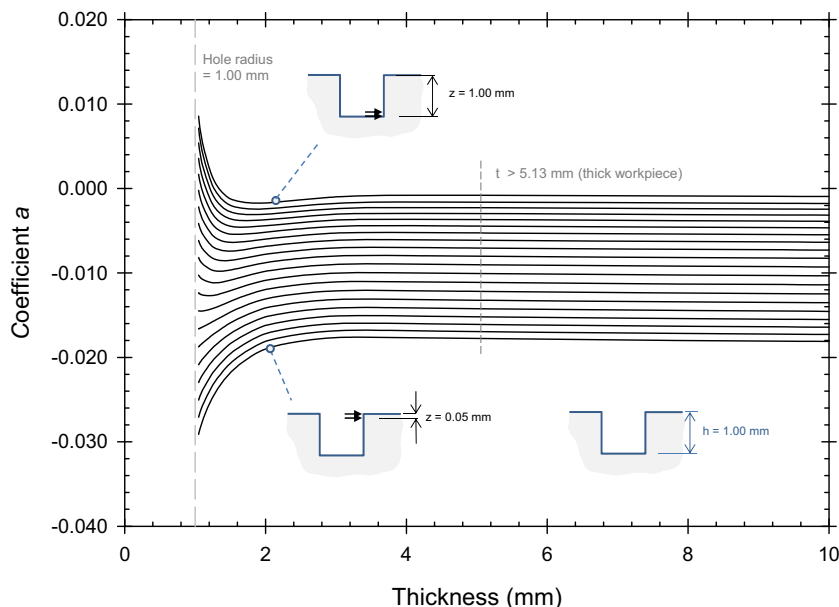
281 tive coefficients are not found for  $z = 1.00$  mm and the  
 282 neighbouring lines. The same curvature is thus found for  
 283 every coefficient in Fig. 8.

284 For both calibration coefficient matrices, it can be con-  
 285 cluded that the thickness limitation of 5.13 mm proposed  
 286 by the standard is too restrictive. A very smooth coefficient  
 287 change is found and the values in the standard are only  
 288 completely invalid for thicknesses less than 3.00 mm.  
 289 Complete matrices for thicknesses 1.05, 1.50 and  
 290 2.00 mm are included in Appendix 1.

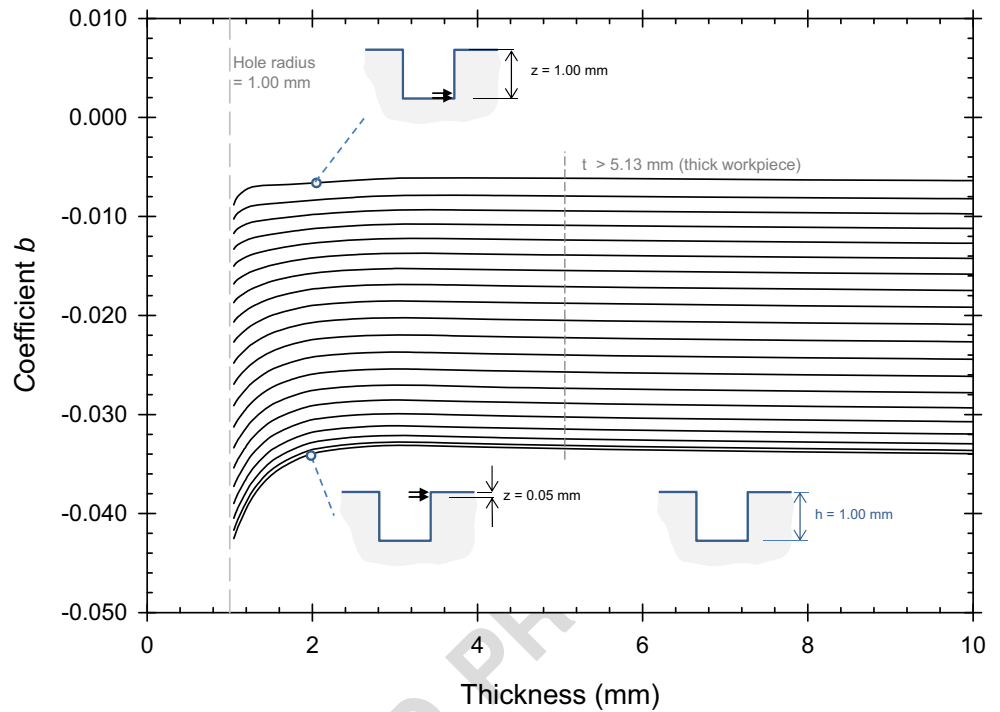
### Effect of the Hole Diameter

291  
 292 The hole diameter is determined by the end mill dimen-  
 293 sion. Although commercial Hole-Drilling devices feature  
 294 a wide range of cutter diameters, ASTM E837 – 13a  
 295 focuses on a general-purpose 1.00 mm radius for the  
 296 drill, so the applicability of the calibration coefficients  
 297 lies in a range of  $\pm 0.06$  mm:  $R_0$  between 0.94 and  
 298 1.06 mm. Figure 9 represents the change in the  $\bar{a}_{ij}$  ma-  
 299 trix, with the coefficient corresponding to  $R_0 = 1.00$  mm  
 300 on the x-axis, and the new coefficient for a different  
 301 radius on the y-axis. Hence, for  $R_0 > 1.00$  mm, the  
 302 points lie below a line with slope one, whereas for  $R_0$   
 303  $< 1.00$  mm, the coefficients lie above this line. This  
 304 result demonstrates how calibration coefficients (or their  
 305 absolute values) decrease with smaller holes and increase  
 306 with larger ones. This behaviour may be attributed to the  
 307 fact that if the ratio  $D_0/D$  is very large, the strain is  
 308 measured close to the edge of the hole, and hence, the

**Fig. 7** Effect of the thickness on the coefficients  $\bar{a}(z, h)$  for hole depth  $h = 1.00$  mm and hole radius  $R_0 = 1.00$  mm. Coefficients correspond to the tenth row of the  $\bar{a}(z, h)$  matrix, with increasing stress depth from bottom ( $z = 0.05$  mm) to top ( $z = 1.00$  mm)



**Fig. 8** Effect of the thickness on the coefficients  $\bar{b}(z, h)$  for hole depth  $h = 1.00$  mm and hole radius  $R_0 = 1.00$  mm. Coefficients correspond to the tenth row of the  $\bar{a}(z, h)$  matrix, with increasing stress depth from bottom ( $z = 0.05$  mm) to top ( $z = 1.00$  mm)



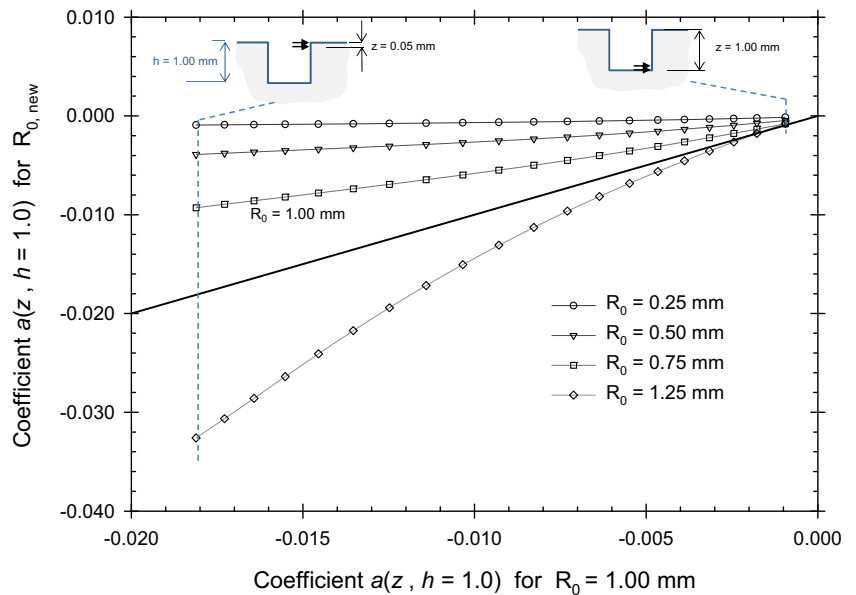
309 deformation must be higher. In a similar way, for  $R_0 <$   
 310 1.00 mm, the results show that the calibration coeffi-  
 311 cients are less sensitive to the hole size. Again, a bulk  
 312 constraint may also have a role. It must be recalled that  
 313 every simulation presented in Figs. 9, 10, 11, and 12 is  
 314 based on a very thick workpiece ( $t = 10.00$  mm).

315 A similar effect is found for the  $\bar{b}_{ij}$  matrix, but an  
 316 even higher increase in the negative deformation is ob-

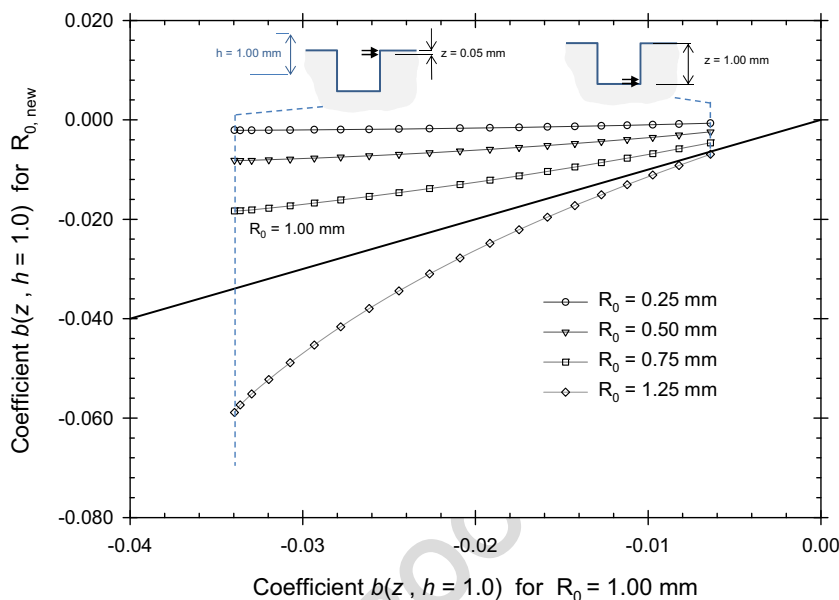
served for large values of  $R_0$ . The curves in Figs. 9 and  
 10 are not easy to fit using a linear regression, even  
 though the proposed correction in ASTM E837 – 13a  
 is as follows [10]:

$$\bar{a}_{ij,R_0} = \left( \frac{R_0}{1.00 \text{ mm}} \right)^2 \bar{a}_{ij,1.00 \text{ mm}} \quad (16)$$

**Fig. 9** Effect of the hole radius on the coefficients  $\bar{a}(z, h)$  for hole depth  $h = 1.00$  mm and thickness  $t = 10.00$  mm



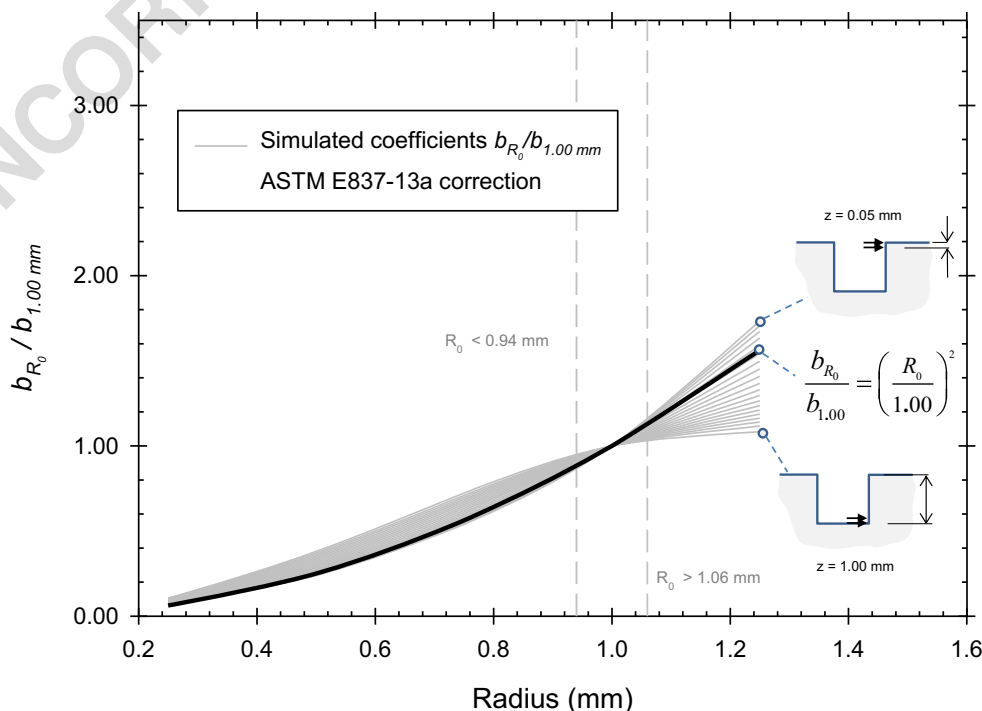
**Fig. 10** The effect of the hole radius on the coefficients  $\bar{b}(z, h)$  for hole depth  $h = 1.00$  mm and thickness  $t = 10.00$  mm



324 With the aim of evaluating the apparent deviation of  
 325 the simulated coefficients from the correction proposed  
 326 in equation (16), the ratio  $\bar{a}_{R_0}/\bar{a}_{1.00 \text{ mm}}$  is graphed against  
 327 the hole radius in Fig. 11. The coefficients obtained  
 328 through FE simulations with different hole sizes are also  
 329 compared with the correction proposed by the standard,  
 330 i.e., equation (16). The latter, which is a parabola, is a  
 331 reasonably good prediction for hole radii near 1.00 mm

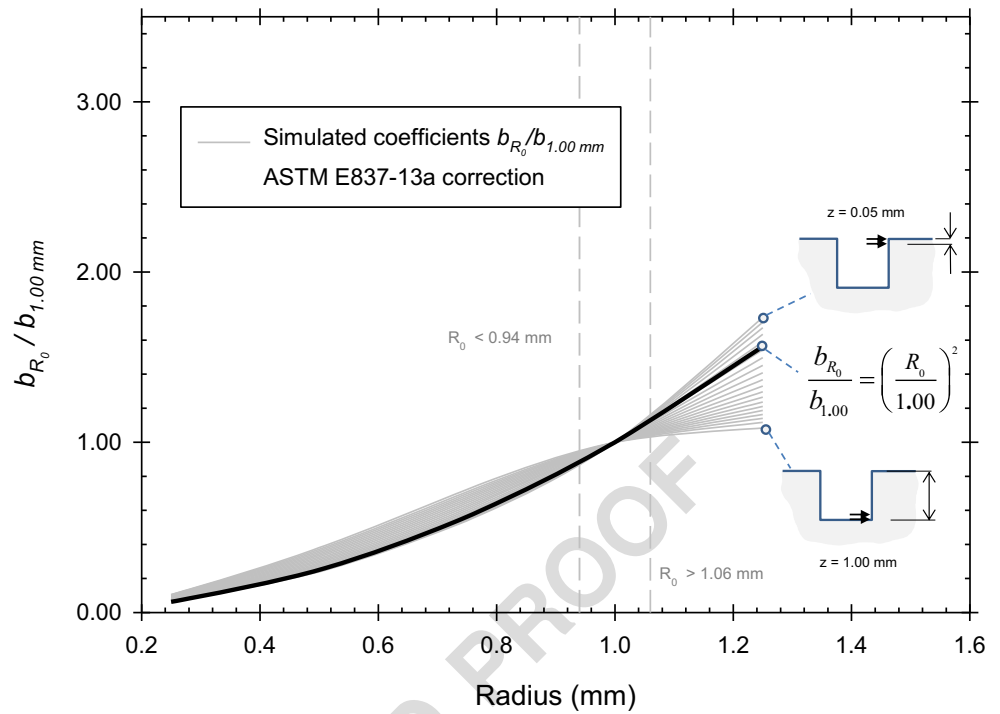
or for coefficients corresponding to a stress applied near 332  
 the free surface, i.e., close to  $z = 0.05$  mm. In contrast, 333  
 coefficients corresponding to a stress applied near the 334  
 bottom of the hole, i.e., close to  $z = 1.00$  mm, are found 335  
 to deviate significantly from the correction given in 336  
 ASTM E837 – 13a. Despite this, the measurements taken 337  
 during drilling at sub-surface depths are highly suscepti- 338  
 ble to measurement errors, so that deviations in numeri- 339

**Fig. 11** Deviation from the hole diameter correction proposed by ASTM E837 – 13a for hole depth  $h = 1.00$  mm and thickness  $t = 10.00$  mm





**Fig. 12** Deviation from the hole diameter correction proposed by ASTM E837 – 13a for hole depth  $h = 1.00$  mm and thickness  $t = 10.00$  mm



340 cal values may be acceptable to a certain degree since  
 341 their contribution to the overall experimental error is less  
 342 than from the measurement errors. The correction by  
 343 diameter considered by the standard, although imprecise,  
 344 can be valid taking into account that regularization tends  
 345 to reduce these effects.

346 Following the same procedure, Fig. 12 shows a comparison  
 347 between the simulated values of  $\bar{b}_{R_0} / \bar{b}_{1.00 \text{ mm}}$  and the correc-  
 348 tion recommended by the standard. Behaviour similar to that  
 349 found in Fig. 11 is obtained. However, note how coefficients  
 350 corresponding to different stress locations are concentrated in  
 351 a narrower spectrum, especially for coefficients close to  $z =$   
 352  $1.00$  mm. Consequently, it can be assumed here that the pre-  
 353 diction given by ASTM E837 – 13a is better for the “b” matrix  
 354 than for the “a” matrix.

355 **Conclusions**

356 A Finite Element procedure was validated as a powerful  
 357 tool for finding coefficient matrices in the Integral  
 358 Method framework as a first step in measuring residual  
 359 stresses using the Hole-Drilling method. Using ANSYS,  
 360 the results were compared with the coefficients given by  
 361 ASTM E837 – 13a; they were in very good agreement  
 362 for type-B rosettes. It can be deduced from this

correspondence that the present simulations are valid  
 for thin strain gauges. In future extensions, the gauge  
 width will be included as an influential parameter.

In addition, the limits on the thickness and hole radius  
 were studied and discussed. The definition of a “thick”  
 workpiece in ASTM E837 – 13a was attributed to the  
 critical change in the calibration coefficients for small  
 thicknesses, which is physically related to the con-  
 straints. The presented results and the matrices given in  
 Appendix 1 are expected to overcome this limitation to  
 experimentally determine the non-uniform residual stress  
 distribution in a thin workpiece. The influence of the  
 hole diameter was also analysed with a focus on a critical  
 revision of the correction proposed in ASTM E837 –  
 13a. It was concluded that the adjustment for a hole  
 diameter different than  $2.00$  mm used by the standard  
 might be improved because the coefficients found in  
 the FE simulations deviated significantly from this cor-  
 rection. The observed deviation is especially important  
 for large hole diameters. As a general conclusion, a par-  
 ticular matrix must be obtained through the exposed au-  
 tomated numerical procedure when the analysed work-  
 piece is too thin ( $t < 3.00$  mm) or when the hole diameter  
 is relatively large ( $D_0 > 1.1$  mm).

**Acknowledgments** The authors gratefully acknowledge financial support  
 from the Junta of Castile and Leon through grant no. BU053U16.

389 **Appendix 1: Matrices for Small Thickness**

390 All calibration matrices correspond to a strain rosette diameter

391  $D = 5.13$  mm and a hole diameter  $D_0 = 2.00$  mm.

392

Q1 **Table 3** Calibration matrix  $\bar{a}_{ij}$  for thickness  $t = 1.05$  mm

Stress depth		Hole depth									
mm	0.05	0.1	0.15	0.2	0.25	0.3	0.35	0.4	0.45	0.5	
0.05	-0.01286										
0.10	-0.01558	-0.01352									
0.15	-0.01789	-0.01614	-0.01379								
0.20	-0.01998	-0.01819	-0.01628	-0.01373							
0.25	-0.02181	-0.02001	-0.01810	-0.01605	-0.01338						
0.30	-0.02339	-0.02156	-0.01967	-0.01765	-0.01550	-0.01277					
0.35	-0.02472	-0.02286	-0.02095	-0.01898	-0.01689	-0.01468	-0.01196				
0.40	-0.02580	-0.02392	-0.02200	-0.02003	-0.01800	-0.01586	-0.01364	-0.01097			
0.45	-0.02665	-0.02475	-0.02283	-0.02087	-0.01885	-0.01678	-0.01463	-0.01242	-0.00984		
0.50	-0.02729	-0.02539	-0.02346	-0.02150	-0.01949	-0.01745	-0.01537	-0.01324	-0.01107	-0.00862	
0.55	-0.02774	-0.02584	-0.02392	-0.02196	-0.01997	-0.01794	-0.01589	-0.01383	-0.01174	-0.00964	
0.60	-0.02804	-0.02615	-0.02423	-0.02228	-0.02029	-0.01828	-0.01626	-0.01422	-0.01219	-0.01016	
0.65	-0.02822	-0.02633	-0.02442	-0.02248	-0.02051	-0.01851	-0.01649	-0.01448	-0.01248	-0.01050	
0.70	-0.02832	-0.02643	-0.02453	-0.02259	-0.02063	-0.01864	-0.01664	-0.01464	-0.01265	-0.01069	
0.75	-0.02835	-0.02647	-0.02457	-0.02264	-0.02068	-0.01870	-0.01670	-0.01471	-0.01274	-0.01079	
0.80	-0.02836	-0.02649	-0.02459	-0.02265	-0.02069	-0.01871	-0.01672	-0.01473	-0.01275	-0.01081	
0.85	-0.02839	-0.02650	-0.02459	-0.02266	-0.02069	-0.01870	-0.01670	-0.01470	-0.01272	-0.01076	
0.90	-0.02847	-0.02656	-0.02464	-0.02268	-0.02069	-0.01868	-0.01666	-0.01464	-0.01263	-0.01066	
0.95	-0.02867	-0.02673	-0.02476	-0.02276	-0.02073	-0.01868	-0.01662	-0.01456	-0.01252	-0.01050	
1.00	-0.02911	-0.02709	-0.02505	-0.02298	-0.02088	-0.01875	-0.01662	-0.01449	-0.01237	-0.01027	
Stress depth		Hole depth									
mm	0.55	0.60	0.65	0.70	0.75	0.80	0.85	0.90	0.95	1.00	
0.55	-0.00732										
0.60	-0.00814	-0.00597									
0.65	-0.00853	-0.00660	-0.00457								
0.70	-0.00877	-0.00687	-0.00502	-0.00314							
0.75	-0.00888	-0.00700	-0.00518	-0.00340	-0.00166						
0.80	-0.00890	-0.00703	-0.00521	-0.00344	-0.00173	-0.00010					
0.85	-0.00884	-0.00696	-0.00513	-0.00334	-0.00162	0.00004	0.00156				
0.90	-0.00871	-0.00681	-0.00494	-0.00313	-0.00136	0.00035	0.00198	0.00342			
0.95	-0.00851	-0.00655	-0.00464	-0.00276	-0.00093	0.00086	0.00259	0.00423	0.00562		
1.00	-0.00820	-0.00616	-0.00415	-0.00217	-0.00022	0.00169	0.00357	0.00540	0.00714	0.00856	

**Table 4** Calibration matrix  $\bar{D}_{ij}$  for thickness  $t = 1.05$  mm

Stress depth											
Hole depth											
mm	0.05	0.1	0.15	0.2	0.25	0.3	0.35	0.4	0.45	0.5	
0.05	-0.01743										
0.10	-0.02040	-0.01865									
0.15	-0.02304	-0.02174	-0.01932								
0.20	-0.02531	-0.02414	-0.02246	-0.01958							
0.25	-0.02748	-0.02636	-0.02484	-0.02276	-0.01947						
0.30	-0.02956	-0.02845	-0.02698	-0.02504	-0.02257	-0.01903					
0.35	-0.03133	-0.03025	-0.02882	-0.02700	-0.02478	-0.02211	-0.01843				
0.40	-0.03295	-0.03189	-0.03050	-0.02874	-0.02665	-0.02425	-0.02147	-0.01769			
0.45	-0.03435	-0.03332	-0.03196	-0.03026	-0.02824	-0.02599	-0.02347	-0.02059	-0.01681		
0.50	-0.03562	-0.03462	-0.03328	-0.03160	-0.02965	-0.02750	-0.02515	-0.02254	-0.01964	-0.01588	
0.55	-0.03677	-0.03579	-0.03447	-0.03282	-0.03091	-0.02882	-0.02654	-0.02409	-0.02144	-0.01853	
0.60	-0.03780	-0.03684	-0.03554	-0.03391	-0.03203	-0.02997	-0.02775	-0.02540	-0.02291	-0.02027	
0.65	-0.03868	-0.03774	-0.03645	-0.03484	-0.03298	-0.03096	-0.02880	-0.02650	-0.02411	-0.02162	
0.70	-0.03948	-0.03856	-0.03729	-0.03570	-0.03385	-0.03184	-0.02970	-0.02745	-0.02512	-0.02273	
0.75	-0.04013	-0.03924	-0.03799	-0.03642	-0.03460	-0.03261	-0.03051	-0.02829	-0.02600	-0.02367	
0.80	-0.04075	-0.03988	-0.03865	-0.03708	-0.03527	-0.03330	-0.03122	-0.02903	-0.02677	-0.02448	
0.85	-0.04131	-0.04041	-0.03915	-0.03760	-0.03585	-0.03391	-0.03183	-0.02966	-0.02743	-0.02519	
0.90	-0.04178	-0.04092	-0.03966	-0.03813	-0.03636	-0.03441	-0.03235	-0.03021	-0.02802	-0.02581	
0.95	-0.04216	-0.04133	-0.04010	-0.03856	-0.03682	-0.03492	-0.03289	-0.03077	-0.02860	-0.02641	
1.00	-0.04254	-0.04169	-0.04047	-0.03898	-0.03727	-0.03539	-0.03337	-0.03126	-0.02911	-0.02695	
Stress depth											
Hole depth											
mm	0.55	0.60	0.65	0.70	0.75	0.80	0.85	0.90	0.95	1.00	
0.55	-0.01488										
0.60	-0.01743	-0.01390									
0.65	-0.01903	-0.01629	-0.01292								
0.70	-0.02029	-0.01780	-0.01517	-0.01201							
0.75	-0.02132	-0.01896	-0.01657	-0.01409	-0.01114						
0.80	-0.02219	-0.01991	-0.01764	-0.01539	-0.01309	-0.01039					
0.85	-0.02294	-0.02073	-0.01854	-0.01640	-0.01428	-0.01214	-0.00969				
0.90	-0.02361	-0.02144	-0.01932	-0.01725	-0.01524	-0.01328	-0.01134	-0.00921			
0.95	-0.02423	-0.02209	-0.02000	-0.01799	-0.01604	-0.01417	-0.01239	-0.01068	-0.00885		
1.00	-0.02480	-0.02270	-0.02066	-0.01870	-0.01681	-0.01502	-0.01331	-0.01172	-0.01026	-0.00882	

**Table 5** Calibration matrix  $\bar{a}_{ij}$  for thickness  $t = 1.50$  mm

Stress depth											
Hole depth											
mm	0.05	0.1	0.15	0.2	0.25	0.3	0.35	0.4	0.45	0.5	
0.05	-0.00913										
0.10	-0.01101	-0.00968									
0.15	-0.01258	-0.01154	-0.00995								
0.20	-0.01402	-0.01296	-0.01178	-0.01000							
0.25	-0.01530	-0.01425	-0.01310	-0.01178	-0.00986						
0.30	-0.01644	-0.01538	-0.01426	-0.01299	-0.01157	-0.00956					
0.35	-0.01744	-0.01637	-0.01525	-0.01404	-0.01268	-0.01118	-0.00915				
0.40	-0.01831	-0.01723	-0.01610	-0.01490	-0.01361	-0.01219	-0.01066	-0.00864			
0.45	-0.01905	-0.01796	-0.01683	-0.01564	-0.01437	-0.01302	-0.01157	-0.01004	-0.00806		
0.50	-0.01967	-0.01857	-0.01744	-0.01625	-0.01500	-0.01368	-0.01230	-0.01085	-0.00934	-0.00744	
0.55	-0.02019	-0.01908	-0.01794	-0.01676	-0.01551	-0.01422	-0.01287	-0.01149	-0.01006	-0.00859	
0.60	-0.02060	-0.01949	-0.01836	-0.01717	-0.01593	-0.01464	-0.01332	-0.01197	-0.01061	-0.00922	
0.65	-0.02094	-0.01982	-0.01869	-0.01750	-0.01627	-0.01499	-0.01368	-0.01235	-0.01102	-0.00970	
0.70	-0.02119	-0.02008	-0.01894	-0.01776	-0.01653	-0.01526	-0.01396	-0.01265	-0.01134	-0.01004	
0.75	-0.02139	-0.02028	-0.01914	-0.01796	-0.01673	-0.01546	-0.01417	-0.01287	-0.01158	-0.01030	
0.80	-0.02154	-0.02043	-0.01929	-0.01811	-0.01688	-0.01562	-0.01433	-0.01304	-0.01176	-0.01050	
0.85	-0.02164	-0.02053	-0.01939	-0.01822	-0.01699	-0.01573	-0.01445	-0.01316	-0.01189	-0.01064	
0.90	-0.02171	-0.02060	-0.01947	-0.01829	-0.01707	-0.01581	-0.01454	-0.01325	-0.01198	-0.01074	
0.95	-0.02175	-0.02065	-0.01951	-0.01834	-0.01712	-0.01587	-0.01459	-0.01331	-0.01204	-0.01081	
1.00	-0.02177	-0.02067	-0.01954	-0.01837	-0.01715	-0.01590	-0.01462	-0.01335	-0.01208	-0.01085	
Stress depth											
Hole depth											
mm	0.55	0.60	0.65	0.70	0.75	0.80	0.85	0.90	0.95	1.00	
0.55	-0.00679										
0.60	-0.00781	-0.00613									
0.65	-0.00836	-0.00703	-0.00548								
0.70	-0.00877	-0.00750	-0.00626	-0.00483							
0.75	-0.00906	-0.00785	-0.00666	-0.00549	-0.00419						
0.80	-0.00927	-0.00808	-0.00694	-0.00583	-0.00475	-0.00357					
0.85	-0.00942	-0.00826	-0.00714	-0.00606	-0.00503	-0.00404	-0.00297				
0.90	-0.00953	-0.00838	-0.00727	-0.00622	-0.00522	-0.00426	-0.00334	-0.00238			
0.95	-0.00961	-0.00846	-0.00736	-0.00632	-0.00533	-0.00440	-0.00351	-0.00267	-0.00180		
1.00	-0.00965	-0.00851	-0.00741	-0.00638	-0.00540	-0.00448	-0.00361	-0.00279	-0.00200	-0.00123	

**Table 6** Calibration matrix  $\bar{D}_{ij}$  for thickness  $t = 1.50$  mm

Stress depth											
Hole depth											
mm	0.05	0.1	0.15	0.2	0.25	0.3	0.35	0.4	0.45	0.5	
0.05	-0.01508										
0.10	-0.01759	-0.01625									
0.15	-0.01965	-0.01874	-0.01690								
0.20	-0.02172	-0.02092	-0.01967	-0.01728							
0.25	-0.02365	-0.02287	-0.02167	-0.01998	-0.01725						
0.30	-0.02531	-0.02457	-0.02350	-0.02199	-0.01996	-0.01696					
0.35	-0.02682	-0.02611	-0.02509	-0.02370	-0.02191	-0.01972	-0.01653				
0.40	-0.02819	-0.02749	-0.02650	-0.02516	-0.02350	-0.02153	-0.01920	-0.01593			
0.45	-0.02944	-0.02875	-0.02775	-0.02646	-0.02490	-0.02307	-0.02095	-0.01850	-0.01524		
0.50	-0.03047	-0.02982	-0.02887	-0.02760	-0.02607	-0.02434	-0.02240	-0.02022	-0.01773	-0.01446	
0.55	-0.03143	-0.03080	-0.02986	-0.02862	-0.02713	-0.02545	-0.02359	-0.02156	-0.01932	-0.01683	
0.60	-0.03227	-0.03166	-0.03074	-0.02951	-0.02805	-0.02641	-0.02461	-0.02266	-0.02057	-0.01832	
0.65	-0.03303	-0.03242	-0.03151	-0.03031	-0.02885	-0.02724	-0.02548	-0.02359	-0.02158	-0.01948	
0.70	-0.03367	-0.03308	-0.03218	-0.03099	-0.02955	-0.02796	-0.02623	-0.02437	-0.02242	-0.02040	
0.75	-0.03424	-0.03366	-0.03277	-0.03158	-0.03016	-0.02859	-0.02688	-0.02505	-0.02314	-0.02118	
0.80	-0.03472	-0.03416	-0.03328	-0.03210	-0.03069	-0.02913	-0.02744	-0.02563	-0.02375	-0.02183	
0.85	-0.03515	-0.03460	-0.03373	-0.03256	-0.03116	-0.02961	-0.02793	-0.02614	-0.02429	-0.02239	
0.90	-0.03552	-0.03498	-0.03412	-0.03296	-0.03156	-0.03002	-0.02836	-0.02658	-0.02474	-0.02287	
0.95	-0.03584	-0.03531	-0.03446	-0.03330	-0.03192	-0.03038	-0.02873	-0.02696	-0.02514	-0.02328	
1.00	-0.03612	-0.03559	-0.03475	-0.03360	-0.03222	-0.03069	-0.02905	-0.02729	-0.02548	-0.02364	
Stress depth											
Hole depth											
mm	0.55	0.60	0.65	0.70	0.75	0.80	0.85	0.90	0.95	1.00	
0.55	-0.01364										
0.60	-0.01587	-0.01280									
0.65	-0.01727	-0.01491	-0.01196								
0.70	-0.01833	-0.01618	-0.01391	-0.01112							
0.75	-0.01919	-0.01718	-0.01512	-0.01296	-0.01033						
0.80	-0.01989	-0.01796	-0.01602	-0.01406	-0.01202	-0.00954					
0.85	-0.02050	-0.01862	-0.01676	-0.01491	-0.01306	-0.01114	-0.00881				
0.90	-0.02100	-0.01915	-0.01734	-0.01557	-0.01382	-0.01208	-0.01028	-0.00811			
0.95	-0.02143	-0.01962	-0.01784	-0.01611	-0.01443	-0.01279	-0.01116	-0.00947	-0.00746		
1.00	-0.02180	-0.02001	-0.01826	-0.01657	-0.01493	-0.01335	-0.01180	-0.01027	-0.00871	-0.00684	

**Table 7** Calibration matrix  $\bar{a}_{ij}$  for thickness  $t = 2.00$  mm

Stress depth											
Hole depth											
mm	0.05	0.1	0.15	0.2	0.25	0.3	0.35	0.4	0.45	0.5	
0.05	-0.00751										
0.10	-0.00909	-0.00800									
0.15	-0.01041	-0.00960	-0.00825								
0.20	-0.01162	-0.01080	-0.00985	-0.00829							
0.25	-0.01271	-0.01190	-0.01097	-0.00987	-0.00816						
0.30	-0.01369	-0.01286	-0.01197	-0.01092	-0.00971	-0.00789					
0.35	-0.01455	-0.01372	-0.01283	-0.01184	-0.01069	-0.00939	-0.00753				
0.40	-0.01532	-0.01447	-0.01358	-0.01261	-0.01153	-0.01031	-0.00896	-0.00709			
0.45	-0.01598	-0.01513	-0.01424	-0.01327	-0.01221	-0.01107	-0.00980	-0.00844	-0.00660		
0.50	-0.01656	-0.01569	-0.01480	-0.01383	-0.01280	-0.01168	-0.01050	-0.00921	-0.00786	-0.00608	
0.55	-0.01705	-0.01618	-0.01528	-0.01432	-0.01329	-0.01219	-0.01104	-0.00984	-0.00857	-0.00725	
0.60	-0.01746	-0.01659	-0.01568	-0.01472	-0.01370	-0.01261	-0.01148	-0.01032	-0.00912	-0.00789	
0.65	-0.01781	-0.01693	-0.01602	-0.01506	-0.01404	-0.01296	-0.01185	-0.01071	-0.00955	-0.00838	
0.70	-0.01809	-0.01721	-0.01630	-0.01534	-0.01432	-0.01325	-0.01215	-0.01102	-0.00988	-0.00875	
0.75	-0.01833	-0.01745	-0.01654	-0.01557	-0.01456	-0.01349	-0.01239	-0.01127	-0.01015	-0.00905	
0.80	-0.01852	-0.01764	-0.01673	-0.01576	-0.01475	-0.01368	-0.01259	-0.01148	-0.01037	-0.00927	
0.85	-0.01868	-0.01779	-0.01688	-0.01592	-0.01490	-0.01384	-0.01275	-0.01164	-0.01054	-0.00946	
0.90	-0.01880	-0.01792	-0.01700	-0.01604	-0.01502	-0.01396	-0.01287	-0.01177	-0.01067	-0.00960	
0.95	-0.01890	-0.01801	-0.01710	-0.01613	-0.01512	-0.01406	-0.01297	-0.01187	-0.01078	-0.00971	
1.00	-0.01897	-0.01809	-0.01717	-0.01621	-0.01519	-0.01414	-0.01305	-0.01195	-0.01086	-0.00980	
Stress depth											
Hole depth											
mm	0.55	0.60	0.65	0.70	0.75	0.80	0.85	0.90	0.95	1.00	
0.55	-0.00556										
0.60	-0.00662	-0.00503									
0.65	-0.00719	-0.00600	-0.00452								
0.70	-0.00763	-0.00651	-0.00539	-0.00403							
0.75	-0.00796	-0.00690	-0.00585	-0.00481	-0.00357						
0.80	-0.00821	-0.00718	-0.00618	-0.00521	-0.00426	-0.00314					
0.85	-0.00840	-0.00739	-0.00643	-0.00551	-0.00461	-0.00375	-0.00274				
0.90	-0.00856	-0.00756	-0.00661	-0.00571	-0.00487	-0.00405	-0.00327	-0.00237			
0.95	-0.00868	-0.00769	-0.00675	-0.00587	-0.00504	-0.00427	-0.00353	-0.00283	-0.00203		
1.00	-0.00877	-0.00778	-0.00686	-0.00599	-0.00517	-0.00442	-0.00371	-0.00305	-0.00242	-0.00172	

**Table 8** Calibration matrix  $\bar{D}_{ij}$  for thickness  $t = 2.00$  mm

Stress depth											
Hole depth											
mm	0.05	0.1	0.15	0.2	0.25	0.3	0.35	0.4	0.45	0.5	
0.05	-0.01414										
0.10	-0.01653	-0.01532									
0.15	-0.01846	-0.01768	-0.01598								
0.20	-0.02043	-0.01976	-0.01864	-0.01638							
0.25	-0.02226	-0.02161	-0.02055	-0.01898	-0.01637						
0.30	-0.02384	-0.02323	-0.02230	-0.02091	-0.01899	-0.01611					
0.35	-0.02526	-0.02469	-0.02380	-0.02253	-0.02086	-0.01878	-0.01570				
0.40	-0.02654	-0.02598	-0.02512	-0.02391	-0.02237	-0.02053	-0.01831	-0.01514			
0.45	-0.02773	-0.02717	-0.02631	-0.02515	-0.02371	-0.02199	-0.01999	-0.01765	-0.01448		
0.50	-0.02870	-0.02818	-0.02736	-0.02623	-0.02482	-0.02320	-0.02138	-0.01930	-0.01691	-0.01374	
0.55	-0.02960	-0.02910	-0.02829	-0.02718	-0.02581	-0.02425	-0.02250	-0.02058	-0.01844	-0.01605	
0.60	-0.03040	-0.02991	-0.02912	-0.02803	-0.02668	-0.02516	-0.02347	-0.02162	-0.01963	-0.01748	
0.65	-0.03109	-0.03062	-0.02984	-0.02876	-0.02743	-0.02593	-0.02429	-0.02250	-0.02059	-0.01859	
0.70	-0.03168	-0.03123	-0.03046	-0.02939	-0.02808	-0.02660	-0.02499	-0.02324	-0.02138	-0.01946	
0.75	-0.03221	-0.03176	-0.03101	-0.02995	-0.02864	-0.02719	-0.02559	-0.02387	-0.02206	-0.02019	
0.80	-0.03266	-0.03222	-0.03148	-0.03043	-0.02913	-0.02769	-0.02611	-0.02441	-0.02263	-0.02080	
0.85	-0.03305	-0.03262	-0.03188	-0.03084	-0.02956	-0.02812	-0.02655	-0.02487	-0.02311	-0.02131	
0.90	-0.03339	-0.03297	-0.03224	-0.03120	-0.02992	-0.02850	-0.02694	-0.02527	-0.02353	-0.02175	
0.95	-0.03369	-0.03328	-0.03255	-0.03152	-0.03025	-0.02883	-0.02728	-0.02561	-0.02388	-0.02212	
1.00	-0.03394	-0.03353	-0.03281	-0.03178	-0.03052	-0.02910	-0.02756	-0.02591	-0.02419	-0.02244	
Stress depth											
Hole depth											
mm	0.55	0.60	0.65	0.70	0.75	0.80	0.85	0.90	0.95	1.00	
0.55	-0.01297										
0.60	-0.01513	-0.01216									
0.65	-0.01647	-0.01420	-0.01137								
0.70	-0.01749	-0.01544	-0.01327	-0.01058							
0.75	-0.01830	-0.01638	-0.01442	-0.01236	-0.00984						
0.80	-0.01896	-0.01712	-0.01527	-0.01341	-0.01147	-0.00911					
0.85	-0.01951	-0.01772	-0.01595	-0.01420	-0.01244	-0.01063	-0.00843				
0.90	-0.01997	-0.01821	-0.01649	-0.01481	-0.01316	-0.01151	-0.00982	-0.00778			
0.95	-0.02036	-0.01864	-0.01695	-0.01531	-0.01372	-0.01217	-0.01064	-0.00907	-0.00718		
1.00	-0.02070	-0.01899	-0.01732	-0.01572	-0.01417	-0.01268	-0.01123	-0.00981	-0.00835	-0.00662	

393

**References**

394  
395  
396  
397  
398  
399  
400  
401  
402  
403  
404

1. Webster GA, Ezeilo AN (2001) Residual stress distributions and their influence on fatigue lifetimes. *Int J Fatigue* 23:375–383. [https://doi.org/10.1016/S0142-1123\(01\)00133-5](https://doi.org/10.1016/S0142-1123(01)00133-5)
2. Barsoum Z (2008) Residual stress analysis and fatigue of multi-pass welded tubular structures. *Eng Fail Anal* 15(7):863–874. <https://doi.org/10.1016/j.engfailanal.2007.11.016>
3. Moen CD, Igusa T, Schafer BW (2008) Prediction of residual stresses and strains in cold-formed steel members. *Thin-Walled Struct* 46(11): 1274–1289. <https://doi.org/10.1016/j.tws.2008.02.002>
4. Heinilä S, Marquis GB, Björk T (2008) Observations on fatigue crack paths in the corners of cold-formed high-strength steel tubes.

- Eng Fract Mech 75(3):833–844. <https://doi.org/10.1016/j.engfracmech.2007.01.010> 405
5. Díaz A, Cuesta II, Alegre JM (2017) A methodology for the numerical assessment of autofrettage influence on hydrogen content near a notch in a 4130 steel pressure vessel. *Theor Appl Fract Mech* 92:205–213. <https://doi.org/10.1016/j.tafmec.2017.07.024> 406
6. Blasón S, Rodríguez C, Belzunce J, Suárez C (2017) Fatigue behaviour improvement on notched specimens of two different steels through deep rolling, a surface cold treatment. *Theor Appl Fract Mech* 92:223–228. <https://doi.org/10.1016/j.tafmec.2017.08.003> 407
7. Alegre JM, Bravo P, Preciado M (2007) Fatigue behaviour of an autofrettaged high-pressure vessel for the food industry. *Eng Fail Anal* 14(2):396–407. <https://doi.org/10.1016/j.engfailanal.2006.02.015> 408

409  
410  
411  
412  
413  
414  
415  
416  
417



418 8. Rossini NS, Dassisti M, Benyounis KY, Olabi AG (2012) Methods of measuring residual stresses in components. *Mater Des* 35:572–588. <https://doi.org/10.1016/j.matdes.2011.08.022>

419

420 9. Schajer GS (2010) Advances in hole-drilling residual stress measurements. *Exp Mech* 50(2):159–168. <https://doi.org/10.1007/s11340-009-9228-7>

421

422

423 10. ASTM E837 - 13a (2013) Standard test method for determining residual stresses by the hole-drilling strain-gage method. ASTM International, West Conshohocken

424

425

426 11. Beghini M, Bertini L, Santus C (2010) A procedure for evaluating high residual stresses using the blind hole drilling method, including the effect of plasticity. *J Strain Anal Eng Des* 45(4):301–318. <https://doi.org/10.1243/03093247jsa579>

427

428

429

430 12. Nobre JP, Kommmeier M, Scholtes B (2018) Plasticity effects in the hole-drilling residual stress measurement in peened surfaces. *Exp Mech* 58(2):369–380. <https://doi.org/10.1007/s11340-017-0352-5>

431

432

433 13. Beghini M, Bertini L (1998) Recent advances in the hole drilling method for residual stress measurement. *J Mater Eng Perform* 7(2):163. <https://doi.org/10.1361/105994998770347882>

434

435

436 14. Beghini M, Bertini L, Mori LF (2010) Evaluating non-uniform residual stress by the hole-drilling method with concentric and eccentric holes. Part II: application of the influence functions to the inverse problem. *Strain* 46(4):337–346. <https://doi.org/10.1111/j.1475-1305.2009.00684.x>

437

438

439 15. Schajer GS (1988) Measurement of non-uniform residual stresses using the hole-drilling method. Part I—stress calculation procedures. *J Eng Mater Technol* 110(4):338–343. <https://doi.org/10.1115/1.3226059>

440

441

442 16. Niku-Lari A, Lu J, Flavenot JF (1985) Measurement of residual-stress distribution by the incremental hole-drilling method. *J Mech Work Technol* 11(2):167–188. [https://doi.org/10.1016/0378-3804\(85\)90023-3](https://doi.org/10.1016/0378-3804(85)90023-3)

443

444

445 17. Schajer GS, Prime MB (2006) Use of inverse solutions for residual stress measurements. *J Eng Mater Technol* 128(3):375–382. <https://doi.org/10.1115/1.2204952>

446

447

448

449

450

451 18. Zuccarello B (1999) Optimal calculation steps for the evaluation of residual stress by the incremental hole-drilling method. *Exp Mech* 39(2):117–124. <https://doi.org/10.1007/bf02331114>

452

453 19. Schajer GS (2010) Hole-drilling residual stress measurements at 75: origins, advances, opportunities. *Exp Mech* 50(2):245–253. <https://doi.org/10.1007/s11340-009-9285-y>

454

455

456 20. Schuster S, Steinzig M, Gibmeier J (2017) Incremental hole drilling for residual stress analysis of thin walled components with regard to plasticity effects. *Exp Mech* 57(9):1457–1467. <https://doi.org/10.1007/s11340-017-0318-7>

457

458

459 21. Schajer GS (1988) Measurement of non-uniform residual stresses using the hole-drilling method. Part II—practical application of the integral method. *J Eng Mater Technol* 110(4):344–349. <https://doi.org/10.1115/1.3226060>

460

461

462 22. Beghini M, Bertini L (2000) Analytical expressions of the influence functions for accuracy and versatility improvement in the hole-drilling method. *J Strain Anal Eng Des* 35(2):125–135. <https://doi.org/10.1243/0309324001514071>

463

464

465 23. Aoh J-N, Wei C-S (2002) On the improvement of calibration coefficients for hole-drilling integral method: part I—analysis of calibration coefficients obtained by a 3-D FEM model. *J Eng Mater Technol* 124(2):250–258. <https://doi.org/10.1115/1.1416685>

466

467

468 24. Xiao B, Li K, Rong Y (2011) Automatic determination and evaluation of residual stress calibration coefficients for hole-drilling strain gauge integral method. *Strain* 47(s1):e525–e534. <https://doi.org/10.1111/j.1475-1305.2009.00650.x>

469

470

471 25. Zhu X, di Scalea FL (2017) Thermal stress measurement in continuous welded rails using the hole-drilling method. *Exp Mech* 57(1):165–178

472

473

474 26. Schajer GS, Abraham C (2014) Residual stress measurements in finite-thickness materials by hole-drilling. *Exp Mech* 54(9):1515–1522. <https://doi.org/10.1007/s11340-014-9935-6>

475

476

477

478

479

480

481

482

UNCORRECTED PROOF